POLITECNICO MILANO 1863

Autonomous untethered flight of multibody dynamics rotorcraft with cycloidal rotors DETC2018-85152

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Outline

Introduction

- project objectives, studied aircraft, cycloidal rotors
- Modeling
- multibody architecture, control algorithm, optimal tuning

Results

- path following, energy consumption

Objectives

- Develop untethered aircraft models
 - traditional helicopter
 - "Quadricyclogyro"
 - a quad cycloidal rotor aircraft
 - "Heligyro"
 - a compound helicopter w/ cycloidal rotors

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Studied Rotorcraft



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Cycloidal Rotors

- Century old concept
 - airborne:
 - only exist as prototypes
 - water:
 - tug boats and ferries
 - advantages:
 - directional thrust
 - cost. span airspeed
 - allows flight at any planar angle



Follow a Blade over a Single Period

Can pitch harmonically

- about its pivot axis
 - ex: 30 sin(Ω t) degrees
- ...and rotates about rotor's center
- Consequently,
- generates:
 - directional thrust
 - inflow
 - rotation resistance (drag)

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Bo105 Helicopter Multibody Model

- Implemented in MBDyn
 - only rigid elements
 - aerodynamic elements for rotor blades
 - with twist
 - based on lookup-table
- Legend
- == 5 dof fixed, pitch free
 - A 3dof control, 3 dof fixed
 - \downarrow 1dof rotation fixed
 - Ψ 1dof rotation imposed
 - rigid connection

🗌 node

- with intertia
- aerodynamic force
- ↓ viscoelastic rod
- 5 dof fixed, rotation free



Cycloidal Rotor Multibody Model

- Simple arrangement of blades
- rotate about central drum
- controlled harmonic pitching about pivot
- custom 1D inflow model



- 5 dof fixed, pitch free
 - 1dof rotation imposed
 - 5 dof fixed, rotation free



Control Algorithms

Each aircraft controlled by similar PIDs

- dynamically adjust control surfaces
- controller input signal is error at current timestep
- btwn filtered 6D position and requested path
- work in both in translation and rotation
- Helicopter and Heligyro have simple Stability Augment. Syst. (SAS)
- Quadricyclogyro:
- forward/rearward/yaw obtain by controlling blade pitch functions
- roll/pitch obtained by differential L/R or F/A rotor pitch functions
- Heligyro:
- controls **yaw** by differential long. thrust of cycloidal rotors

Tuning the controllers

- Controllers tuned by path error minimization
 - Single-objective Genetic Algorithm (SOGA)
 - 28 to 32 optimization variables
 - using flight time between 2 and 5 minutes
 - minimizing energy consumption
 - *minimizing cabin vibrations*
 - ran with MPI for many-task computing (MTC)

Path-Following Results

• Hover

- comparison btwn requested and flown paths
- Helicopter, Heligyro, Quadricyclogyro



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Path-Following Results

"Trickier" path

- comparison btwn requested and flown paths
- Helicopter, Heligyro, Quadricyclogyro



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Energy Consumption Results

Heligyro

1000

800

600

400

200

0

20

Total power, incl. crude est. of tail (kW)

- requires similar power than trad. helico.
 - more testing and adapted controller may lead to reduced
 - power consumption by:
 - removing any aircraft pitch
 - flying at higher speeds



100

Time (s)

120

Conclusion

- Developed new multibody dynamics models
- totally untethered flight
- tests unconventional configurations
- embeds unconventional rotors such as cycloidal rotors
- Constructed a methodology for control
- optimized through minimization coupled with power reduction
- follows challenging paths w/o predictor
- Heligyro shows good handling properties
- slightly more efficient than traditional helicopter

implementation details provided in paper

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Thank You for your Attention. <u>Questions?</u>

Video (2 min): https://youtu.be/WSIK4t-Rg1Q

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